

# ***Experimental Investigation of the Shuttle Transportation System Composite Overwrapped Pressure Vessels for Stress Rupture Life***

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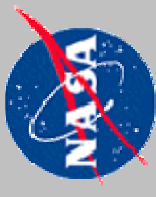
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# *Outline*

- Introduction
- Purpose for Testing
- NASA WSTF COPV Test Program
- NASA WSTF Test Facilities
- COPV Impact Study
- Fluids Compatibility Testing
- Stress Rupture Testing
- COPV Lifing
- Conclusions



# *Introduction*

- Composite overwrapped pressure vessel (COPV).
  - Typically a metallic liner overwrapped with a fiber epoxy matrix
- Weight advantage: Kevlar COPVs saved 752 lb in pressurized storage over an all metal design.



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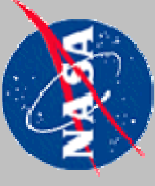
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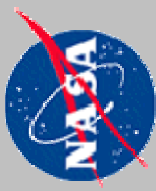
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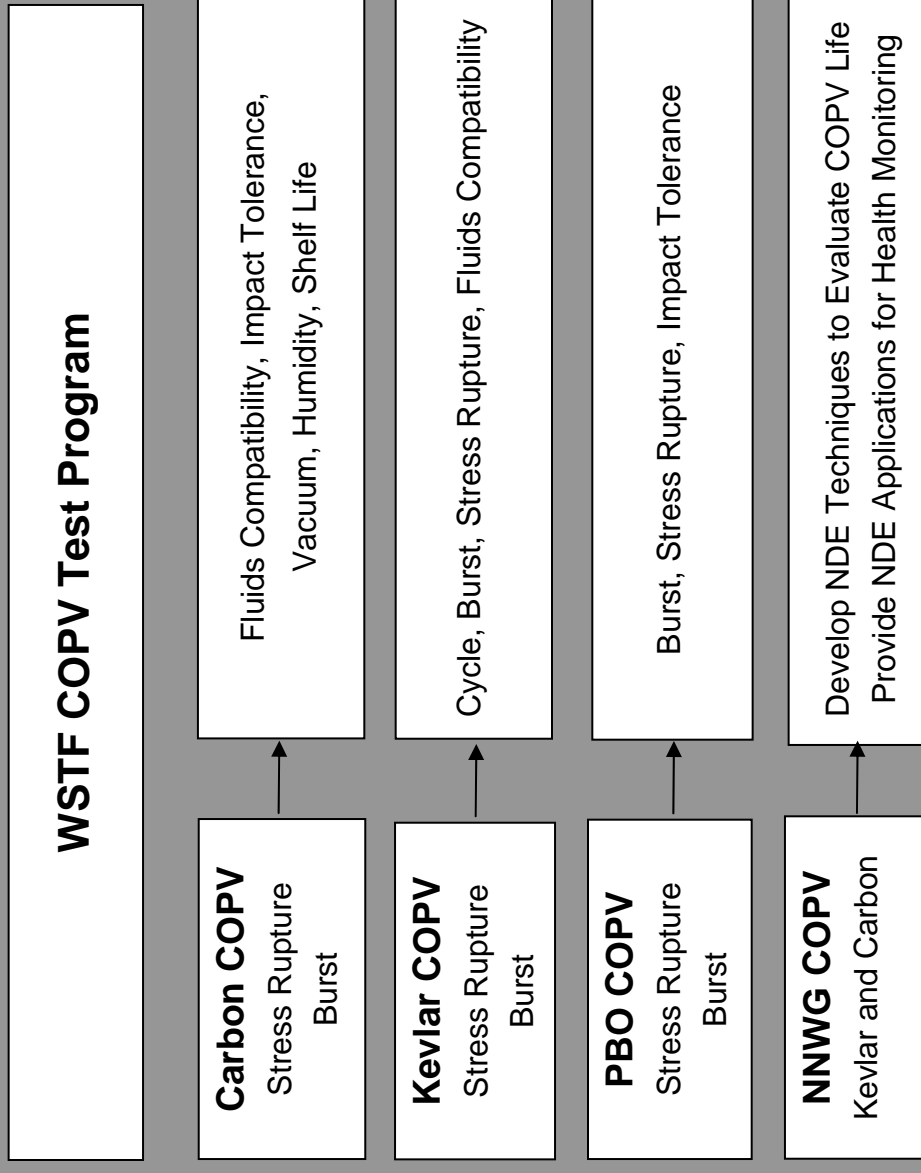
# ***Purpose for Testing***



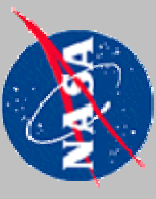
- Safe use of COPVs
- Two failure modes for a COPV
- Leak Before Burst (LBB) more benign failure mode
- Burst Before Leak (BBL) catastrophic failure mode with pressure wave and fragments
  - Stress rupture: sudden failure of the overwrap BBL
  - Composite Damage: impact or cuts can cause BBL
  - Liner Flaw: dump of load into composite can cause BBL
  - Manufacturing: defects can cause BBL



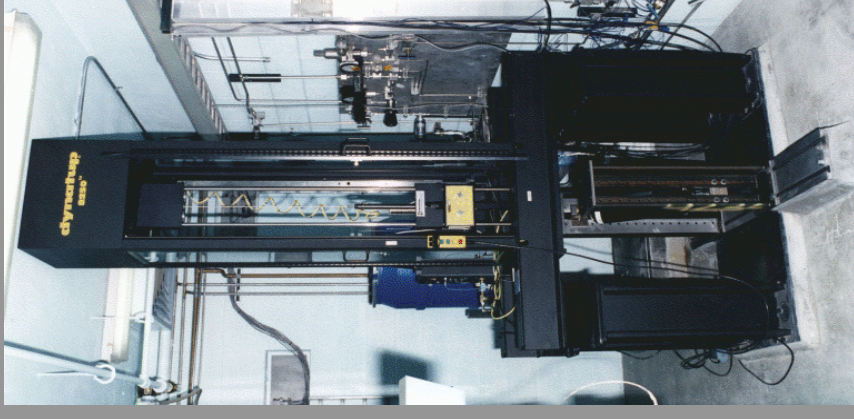
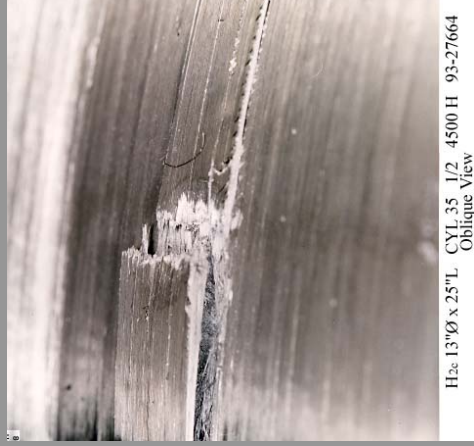
# NASA-WSTF COPV Test Program



# *COPV Impact Study*



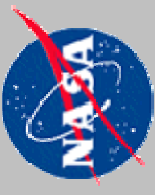
- Understand COPV impact sensitivity
- Establish VDT, CIE
- Data to support impact control plan



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# Fluids Compatibility Testing



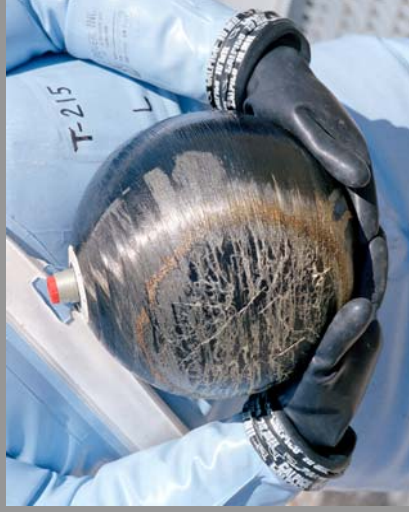
- Effect of fluids on COPV strength<sup>1</sup>
- MMH, NTO, LN<sub>2</sub>, hydrazine, and unsymmetrical dimethylhydrazine
- Carbon not sensitive to fluids
- Kevlar very sensitive to NTO



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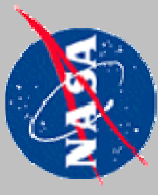
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<sup>1</sup> Details of testing is in "Composite Overwrapped Pressure Vessels NASA/TP-2002-210769



# *Stress Rupture Testing*

- Carbon and PBO Stress Rupture Test
- Thin overwraps similar to propellant tanks



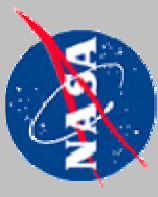
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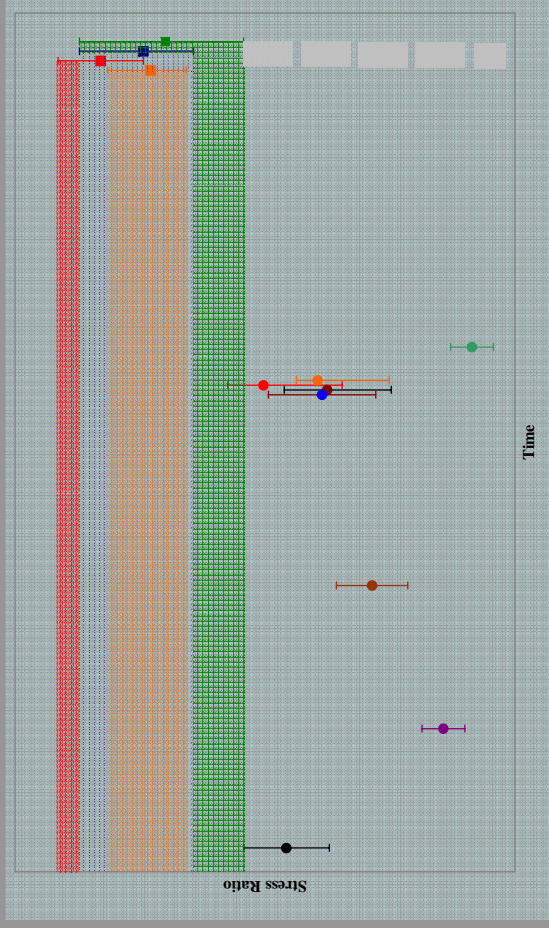


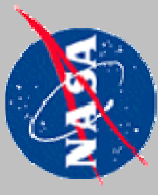


# Stress Rupture Testing

- Flight-qualified configurations
- COPVs in test for 9 years
- Nine impact-damaged COPVs
- Limited statistical population

| Size (in)    | Undamaged COPV<br>Fiber Stress (% Ult) |
|--------------|--|
| 9 Ø          | 62                                     |
| 6.6 Ø X 20 L | 56                                     |
| 13 Ø X 25 L  | 58                                     |
| 10.25 Ø      | 60                                     |



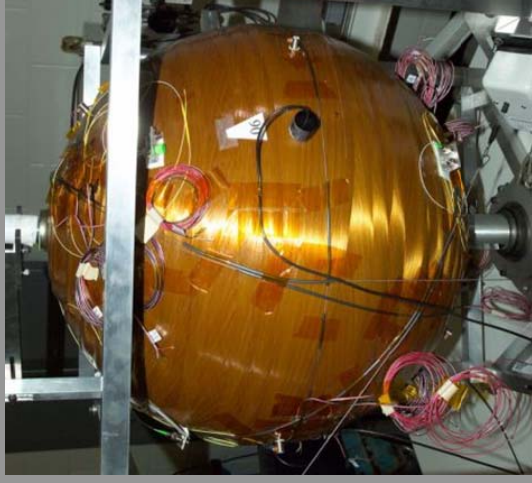


# ***Stress Rupture, Cycle and Burst Testing***

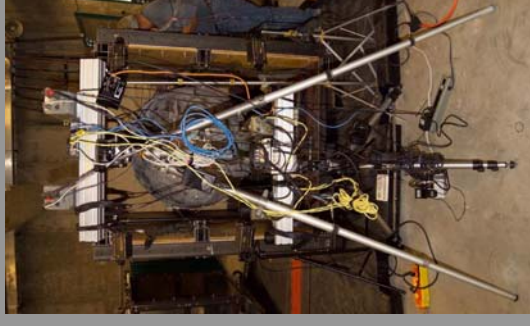
- Thermal control for constant vessel pressure
- Backup power for data acquisition and thermal control
- Flight qualified Kevlar® COPV testing
  - Strain, FOBG, Volume Measurement, DIC, Eddie Current, AE, Raman Spectroscopy, Belly Bands and NDE<sup>1</sup>



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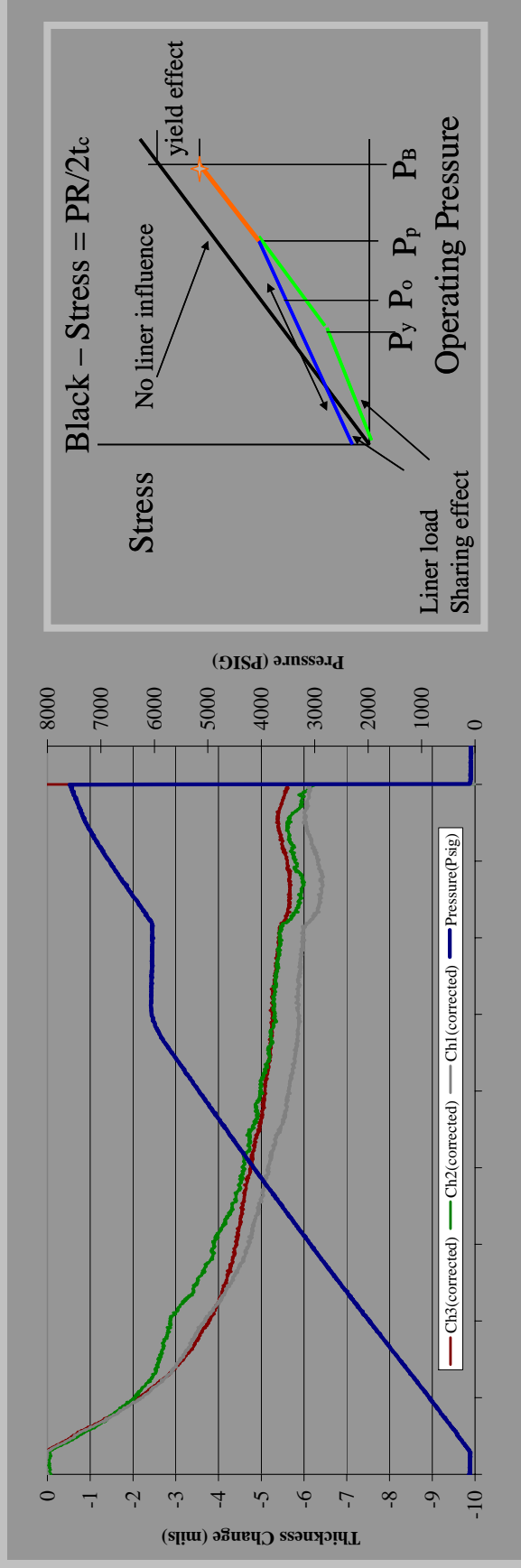


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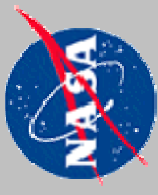
# COPV Lifting

- Establish the amount of liner load sharing<sup>1</sup>
  - Evaluate through the composite compression
  - Liner residual stress after yield point
  - Place time at pressure and stress ratio into model<sup>2</sup>



- <sup>1</sup> Details of how stress ratio calculation is in companion paper by Thesken
- <sup>2</sup> Details of stress rupture lifing is given in a companion paper by Phoenix

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# *Conclusions*

- COPVs are sensitive to composite damage and should have impact control plans
- Propellants and fuels exposure to carbon COPV composite does not immediately affect COPV burst strength
- Limited stress rupture test data for Carbon COPVs (difficulty in predicting stress rupture life)
- Testing to understand COPV mechanical response is important in estimating stress rupture life

# *Questions*

